In the Claims:

Claims 1 to 12 (Canceled).

- 13. (Currently amended) Intake An abradable shroud lining (34) for a gas turbine for the low-wear grazing of components, especially seal fins, seal fins that are movable relative to the intake abradable shroud lining, which consists of a metal foam (28) rigidly connected with at least one carrier (29), characterized in that the or each at least one carrier (29) comprises has openings or bored holes (35) and the metal foam (28) is embodied open-pored, so that both the or each at least one carrier as well as the metal foam are through-flowable in a radial direction of the qas 10 turbine, and the metal foam is bare and directly exposed to 11 the seal fins so that the seal fins graze into the metal 12 foam. 13
- 14. (Currently amended) <u>Intake</u> <u>The abradable shroud</u> lining according to claim 13, characterized in that the metal foam (28) has a stepped contour.
- 15. (Currently amended) Intake The abradable shroud lining according to claim 14, characterized in that the metal foam (28) is rigidly connected with the carrier (29), especially (29) by being glued or soldered thereto.

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- 16. (Currently amended) Intake The abradable shroud lining according to claim 13, characterized in that the metal foam (28) is rigidly connected with the carrier (29), especially (29) by being glued or soldered thereto.
- 17. (New) A gas turbine engine comprising a rotatable rotor with rotor blades, seal fins on radially outer blade tips of the rotor blades, and an abradable shroud lining arranged circumferentially around the blade tips such that the seal fins graze the abradable shroud lining, wherein:

the abradable shroud lining comprises a carrier and an open-pored metal foam that is rigidly connected with the carrier,

the metal foam of the abradable shroud lining is bare and exposed and arranged relative to the rotor so that the seal fins directly graze the metal foam, and

the carrier has holes therein allowing gas communication through the holes and through the open-pored metal foam in a radial gas flow direction as defined with respect to an axis of the gas turbine engine.

[RESPONSE CONTINUES ON NEXT PAGE]